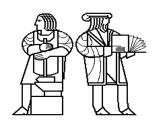
NEXTOR Annual Research Symposium November 14, 1997

Session III Issues for the Future of ATM

Reusable Launch Vehicles James Kuchar, MIT

Integration of Reusable Launch Vehicles (RLVs) into Air Traffic Management



Massachusetts Institute of Technology

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First Annual NEXTOR Research Symposium Washington, DC November 14, 1997 Problem identified by FAA Office of Commercial Space Transportation

Current mode:

- Strategic segregation through Special Use Airspace (SUA)
- Large spatial and temporal buffer
- Based on range safety studies
- Few space operations --> limited impact on air traffic flow

Future:

- Increased frequency of space operations
- Airspace demand conflicts between air and space operators
- Increased diversity of space vehicles (Reusable Launch Vehicles, RLV) Similar to conventional aircraft in some phases of flight
- Need to make most efficient use of airspace

Airspace allocation

- Exclusively to air (SUA not active)
- Exclusively to space (SUA active)
- Integrated / mixed use

Under what conditions could RLVs be integrated into ATM?

- also other unconventional vehicles (e.g., unmanned aerial vehicle)

What would be the economic / flow / safety impacts?

What additional technologies / equipment / procedures are required?

→ Studies needed to determine cost / benefit tradeoff

MIT: Technology (e.g. sensor) and procedural (e.g. human factors) requirements

VPI: Traffic flow modeling and economic analysis

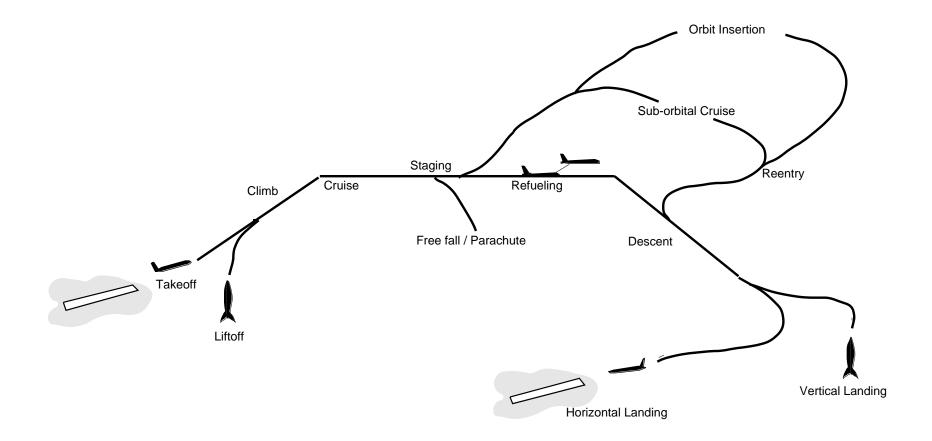
Project phases

Phase I April - September, 1997 Identification of models and issues

Phase II

October, 1997 - September, 1998 Preliminary analysis, recommendations

- Collected data on proposed RLVs
- Determined typical phases of flight / mission profiles
- Developed generalized model of airspace / air traffic / RLV operations
 - Agents
 - Information flow
- Identified 8 potential modes of operation, defined preliminary requirements examples:
 - Continue use of SUA (strategic segregation)
 - Controlled Space Activity Zone (c.f. Class B airspace)
 - Manage RLV as a conventional vehicle



RLV Mission Profiles

| Vehicle | Takeoff | Liftoff | Climb | Cruise | Staging | Refueling | Sub-orbit | Orbit | Reentry | Descent | H Landing | V Landing |
|------------------------|---------|---------|-------|--------|---------|-----------|-----------|-------|---------|---------|-----------|-----------|
| Conventional | | x | x | | x | | x | x | | | | |
| EELV | | x | x | | x | | x | x | | | | |
| Sea Launch | | x | x | | x | | x | x | | | | |
| Shuttle | | x | x | | x | | | x | x | x | x | |
| X-33 | | x | x | | | | x | | x | x | x | |
| Venture Star | | x | x | | | | x | x | x | x | x | |
| DC-Y | | x | x | | | | x | x | x | x | | x |
| Kistler K-1 | | x | x | | x | | x | x | x | x | | x |
| X-37 | | x | x | | x | | | x | x | x | | x |
| Roton | | x | x | | | | x | x | x | x | | x |
| HOTOL | x | | x | x | x | | x | x | x | x | x | |
| Pegasus | x | | x | x | x | | x | x | | x | x | |
| Sänger | x | | x | x | x | | x | x | x | x | x | |
| X-34 | x | | x | x | x | | x | | x | x | x | |
| Pioneer/Pathfin der | x | | X | x | x | X | x | x | X | X | X | |

| NASP x x x | x x x x |
|------------|---------|
|------------|---------|

• Instantaneous Impact Point (IIP)

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Real-time display
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Accuracy (3)
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Along-track: < 100' or 5% of range (whichever is larger) Cross-track: < 100' or 0.5% range (whichever is larger) Telemetered data update rate 20 Hz IIP update rate 10 Hz Maximum destruct delay of 1.5 - 3.5 sec Tracking system reliability 0.999 over 1 hour

- Abort boundary based on safety studies
 - IIP outside abort boundary --> destruct / abort

- Collision with vehicle
 - Sensors, guidance accuracy, geometry, velocity, time, maneuverability
- Wake vortex
- Exhaust / chemical plume / smoke
- Expended stages
- Auxiliary operations: chase planes, weather soundings
- Potential for catastrophic failure
 - departure from planned trajectory
 - explosion / debris

Taken together, uncertainties result in need for large safety buffer -> SUA

Use SUA unless significant negative impact on air traffic occurs

Operational simplicity Most flexibility for RLV operations Safety

Subdividing SUA may alleviate some traffic flow problems (charting issues remain)

Real-time feedback from RLV Operator to ATC could be beneficial Controlled air traffic may enter certain regions of SUA on case-by-case basis Additional controller tools, communications, procedures needed

Non-SUA modes of operation require more research

- Fundamental limitations of current / future ATM concepts
- (e.g., max velocity, vertical rate, excursion from flight plan)

Additional research required to examine feasibility of non-SUA modes of operation Appropriate safety buffer size and duration Equipage and procedural requirements Not clear what the limits of ATC are Ability to manage high speeds / vertical rates Display / procedure / control issues

- Determine relationships between vehicle characteristics and ATC requirements
 - Sensors / tracking
 - Update rate, accuracy
 - Human / automation
 - **Display requirements**